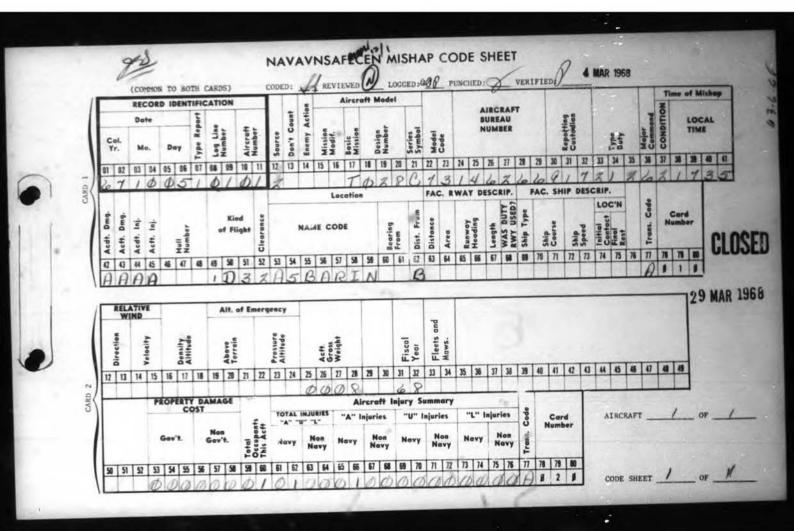
AIRCRAFT ACCIDENT IDENTIFICATION NO.

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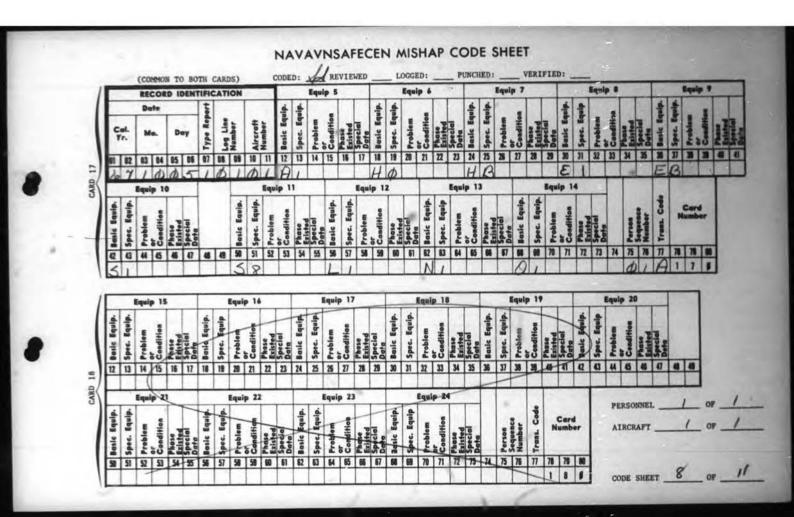
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NAVAVNSAFECEN MISHAP CODE SHEET REVIEWED PUNCHED: LOGGED: _ CODED: 5 (COMMON TO BOTH CARDS) Controlling LSO's Carrier Pess Be RECORD IDENTIFICATION Col. Me. Day 2 17 4 15 16 17 18 18 16 11 Speed Speed Modiff. Line-Up Modiff. Power Position Alv. CARD CLCPD (cost'd) 1 TOUCH-DOWN Card Speed Speed Speed Line-Up Modif. Power Nose Positien A. 12 13 14 15 16 17 18 19 20 21 22 23 24 25 25 77 78 29 30 31 32 33 34 35 38 37 38 28 48 48 42 48 48 48 48 48 48 12 CARD tnj. to Indiv. Abandon A/C Trans. Cade Br. of Service Age Yrs, D.N.A. Cord Rosk/Rate AIRLNAFT Position Status 58 51 52 53 54 55 56 57 58 59 68 51 67 67 65 65 67 58 69 70 71 72 73 74 75 76 77 78 78 78 78 A 1 2 1

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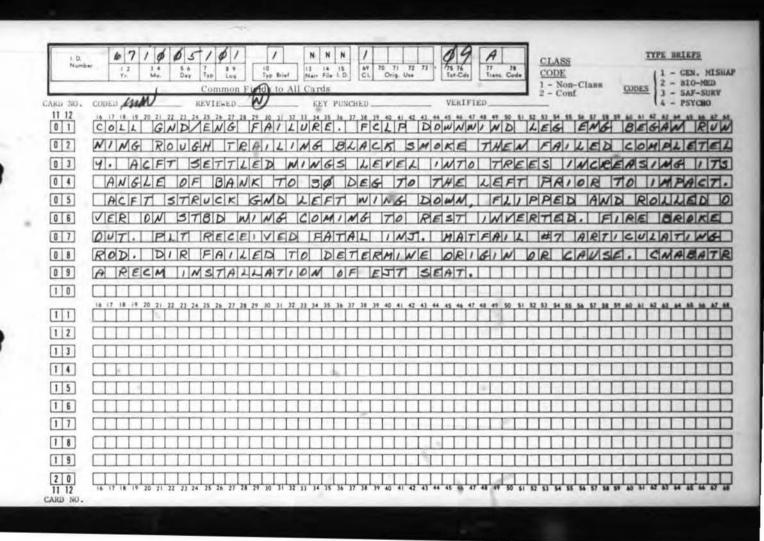
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Date 10-6-6 Dispatch Code Sheets Acft. Accidents Only (Rev. 2-67) Columns Codes Card No. 919 01-11 Record Ident. (common all cards) 12 Source Don't Count 13 **Enemy Action** 14 Aircraft Model 7028015-21 Model Code -7 3 22-23 Aircraft BuNo. 146266 24-29 Reporting Custodian Type Duty 2 33-35 Major Command Time of Day 2/735 37-41 Accident Damage Aircraft Damage 43 Accident Injury 44 Aircraft Injury 45 Hull No. 46-48 Kind of Flight 49-51 Location (Name Code) A 5 B A R I N 53-59 Transaction Code 77 Card No. 0 1 0 78-80 Card No. 929 33-34 Total "A-U-L" Navy Injuries 61-62 Total "A-U-L" Non-Navy Injuries 63-64 Transaction Code Card No. Card No. 939 Primary Accident Type Primary Phase of Operations Transaction Code Card No. Card No. 949 Type Operations Contributing Causes Transaction Code Card No. Card No. 969 Primary Cause Special Data & Conditions Transaction Code Card No.

See reverse side

Card No. 150

No. of Personnel Records Transaction Code Card No.

Card No. 160

P Pilot's Name Status Transaction Code Card No.

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AAR ADMINISTRATIVE CLOSE-OUT FORM

March 29, 1968

MEMORANDUM FOR THE RECORD

 Subj:
 VT-5 (Activity)
 AAR 5-68 concerning T280 (ACFT Model)

 BUNO 146266 occurring (Date)
 0 ccurring (Date)
 10-5-67 (Date)
 pilot (ELLY)

Ref: (a) Code 50 MEMO of 20 Mar 1968

1. In accordance with reference (a), a close-out letter will not be prepared on subject AAR.

Respectfully,

(b) (6)

1

DEPARTMENTAL COMMENTS FOR "CLOSE OUT" LETTER ON ORIGINAL REVIEW

NOTE: 1. Negative report is required.

- 2. Positive comments will be in a format suitable for inclusion in the "close out"
- 3. Attack additional sheets if more space is required.

M&M DEPARTMENT: The PRIMARY CAUSE OF THIS ACCIDENT WAS MATERIAL FAILURE OF THE NUMBER 7 ARTICULATING ROD. NO FAILURE TRENDS PRESENT THIS AREA IN THE 1820-86 ENGINE

ABRO-MED DEPARTMENT: 18/32

COMPLETION SHEET

Action to Correction	to	Action Required	Completed Code/Date
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NOTES: 1. No person other than those assigned to the Records Control Branch will remove any part of this document from the folder.

 Departments will be fully responsible and accountable for documents in their custody until checked back into Records Control Branch.

3. Any department desiring to retain this report longer than five (5) working days must notify Records Control Branch of their need for extension.

S. DATE OF DATE (200)NAVAIREWORKFAC Pensacola 10-11-67 1. ASSEMBLY (Bedet) 4. ASSEMBLY (Serial) S. DATE SCHOVED 9. DEMOVED FROM (Eng Bod) 10. MEMONED FROM (Eng five) 66640 10-9-67 R1820-86A 520721 18. MS. POCK | 18. A(RCHAFT (Badel) SINCE MET LAST GAM 13. DATE LAST 14. LAST DVERMAUL ACTIVITY 17 AIRCRAFT (BIRD) -28-66 NAVAIREWORKFAC Pensacola 146266 4737 IR. OPERATING ACTIVITY 20. REASON FOR REMOVAL AND CHOS AAR 5-68A Accident/Incident Damage 4.b. TRARON FIVE II. FIND INGS N NON-BASIC (MAINT/OPER) cma. BASIC (MEG/DESIGN) F FOREIGN COLECT DI NO DISCREPANCY 430079N3 070 23 to . DESCRIPTION OF FINDINGS (Include mase and part no. of primary part failure) 23. BISCHEPANT PARTS (Part No.) cono. (See below) NAVAIRSYSCOMREP Pensacola Control Nr. R1820-20-68 refers. TO: NATSF PHILA CC: NASC (AIR 4113) COMBIX NASC (AIR 5362) NAVPIANTREPO PATTERSON MASC (AIR 4041) CHNAVNAT NAVAIRSYSCOMREP PNCLA DIRAFIP CINCLANTFLT NASCIECHREP WOODRIDGE DAS NORTON AFB NAVAVNSAFBCBN -**HUMBER** 15.8 NO CNO CNATRA R1820BB607 ar. conclusions JAG CNABATRA TRARON FIVE (See below) M. RECOMENDATIONS None APPLICABLE X PRIDRITY 061818Z OCT 67 REF PNCIA INCORPORATED

of windings

1. Subject engine sustained severe fire and impact damage. The front crankcase was fractured and broken. All accessories and engine components aft of the power case were destroyed by fire.

WEAPONS ENGINEERING DEPT HEAD

14 . DATE

12 October 1967

- 2. Disassembly inspection revealed that the number seven articulated rod had fractured at the knuckle pin strap.
- 3. The number eight rod was fractured approximately five inches below the piston pin boss and the number six rod was slightly bent. The sides of both rods were damaged where contact was made with the fractured end of the number seven rod during operation.
- 4. The number seven cylinder skirt was broken on each side where contacted by the fractured number seven rod.
- 5. The 2045D19 crankcase main rear section to crankcase main front section bolt was broken between cylinders number seven and eight.

PRI DIR 798 12 October 1967

27. CONCLUSIONS:

1. None.

- 2. The number seven rod is considered to be the initial failure. A laboratory analysis of the rod failed to reveal the cause or origin of failure. The fracture surfaces were severely damaged from engine operation subsequent to the rod failure.
- 3. The damage incurred by the number six and eight rods was caused by the failure of the number seven rod. A laboratory analysis of the number eight rod revealed that it had failed from overload.
- 4.-5. Other internal damage resulted from the number seven art rod failure.

. OVERHAUL ACTIVITY		2. REPORT NO.	9. DATE OF 8/1	4. ASSEMBLY NO	MENCLATURE AND	PART NO.	(2	200)	ENGIN
NAVAIREWORKFAC :	the second secon	796	10-10-67	Governor	, Prop.	4G10-3P	2	1	
. ASSEMBLY (Bodel)	6. ASSEMBLY (Ser		7. ASSEMBLY NEW	. DATE HEMOVE	100000000000000000000000000000000000000	race (Sag Bed)	10. REMOVED F	ROS / SAS	Ber j
4010-3	The second second second second	T53-395-1	73030	UNK	R1820		520721	-	
UNK UNK	UNK	UNK.	OL ACTIVITY		UNK	T-28C	PERSONAL TRANSPORTER TO SERVICE AND SERVIC	16266	1000
. OPERATING ACTIVITY	to. FUR -EFR - AAR	The state of the s	29. BEASON FOR AS	-		V/	-		
TRARON FIVE	AAR 5-6	58A	Accident	/Incident	Damage				
0 NO DISCHEPANCY	B MFG/DESH	GN) N	MAINT/OPER!	F	STANASE SOAMA		No. 1	caro.	Im
NASC (AIR NASC (AIR CHATRA CHABATRA NAVAIRSYSC DCASO WIN	A 4105B) 4041) 4113)	NAVAV JAG COMSI C.N.O CHNAV DIRAF CINCI DAS R	TRISAFECEN CX THAT FIF ANTIFLIT FORTON AFB TIVE (AV	SAF OFFICE OF ALIFE C			MENT BULLET HE	CHANGES , 18.8	41C.
. ,		ICIA	061818z	OCT 67		***	INCORPORATED	-	F
3)			WHAPONS	ENGINEERI	NO DEPT	. HEAD	11 Octob	er 1	967

indication of malfunction or material defects.

27. CONCLUSIONS:
Governor is believed to have been operating satisfactorily prior to impact.

NAVAIREWORKFAC P	ensacola	795	10-10-67	Propell		D. Carlotte and Co.	(200)	X
ASSEMBLY (Hedel)	195064	ial)	73030	UNK	R182	0-86A	BL 52		Ser)
1. TOTAL HES 11. HES SINCE SINCE LAST 0/H	IS. DATE LAST	14. LAST OVERN	AUL ACTIVITY		S. NO. PREV	T-28C	lodel) iT.	14626	
8. OPERATING ACTIVITY	18. FUR -EFR - AA		20. REASON FOR BE		6				
TRARON FIVE	AAR-5-	-68A	Acciden	t/Inciden	t Damag	e 4.b.			
0 NO DISCREPANCY	BASIC (MFG/DESI	gn) N	X (MAINT/OPER)	F	OREIGN OBJ	18.44	He,)	COND.	2 Cont
. DESCRIPTION OF FINDINGS (Za	clade near and par	t so, of prisory	part (silers)			23. D/SCR	PANT PARTS (PA	rt Ha. j	CONG
TO: NATSF PHILA CC: NASC (AIR 4 NASC (AIR 4 NASC (AIR 4 CNATRA CNABATRA	105B) 1041) 113)	JAG COMS C.N.	0.		S	Ba. PERTIN	MENT BULLETINS	CHANGES.	ETC
NAVAIRSYSCO NAVPLANTREP		100	VMAT				HUBBER	VES	NO
a. All discrepa attributed to im	ncies are	CINC DAS TRAF	CLANTFLT NORTON AFB CON FIVE (A CON FIVE (A	VISAFE OF		, =			
			T RET	CARA	11 11			-	-
None.	/1-				100	25.		-	-
PEOUESTED BY	SOOTRUP I	NCIA	061818Z	OCT 67		GREE	INCORPORATE	D O	2

26. FINDINGS:

- a. No. 1 blade was bent approximately 40 degrees aft at 14 inches from hub.
- b. Disassembly revealed that the two drive pins and two screws securing the blade bushing to No. 1 blade were sheared.
- c. Impression on teeth of No. 1 blade gear segment made by mating teeth of rotating cam indicated that propeller was in full low pitch position at time of impact. No indication of malfunction or material defects prior to impact was found.

MAYA TREWORKEAC	Persucula	795	10-10-67	Paragen 2.1	Lett 43 D5	MAT NO. 50-321	(30	0)	Ending.
L'ASSEMLY (Bedel)	4. ASSEMBLY (Bert	(el)	7. ASSEMBLY MFR	. DATE REMOVED	D S. HEMOVES	o room (Eng Hed)		C. C. C. L. C.	Ser)
- In the same	1,95064	V	73030	USEK		320-86A	BL 5207	- AND THE REAL PROPERTY.	2000
11. TOTAL HRS 12. HRS SINCE NEW CAST O/N	*	14. LAST OVERH	AUL ACTIVITY		O/H'S	T-280	2.3	16266	
18. OPERATING ACTIVITY	10. FUR -EFR - AAR	222 21 72 22 22 2	29. REASON FOR RE	The state of the s					
TRARON FIVE	AAR-5-6	SA	A.out.Space	t/Inetileest	Decumen				
O NO DISCREPANCY	B BASIC B DISCREPANO	GN) N	MON-BASIC (MAINT/OPER)	FI:	FOREIGN OBJE		ARY PART PAILURE	com.	2 that
	include near and part	t so, of pricery	part failure)			#5. DISCE	MEPANT PARTS (PART	He.)	com.
NASC (AIR NASC (AIR CNATRA CNABATRA NAVAIRSYSC	4105B) 4041)	JAG COMSTI C.N.O A CHNAVI ON DIRAF	MAT TP	OR LOCKS		24. PERTI	INEMT BULLETING, C REPORTED NUMBER	CHANGES,	ETC.,
17. CONCLUSIONS			ANTFLT				19		
a. All discrep attributed to i M. RECOMMENDATIONS None.		e. TRARO	ORTON AFB ON FIVE (AVE ON FIVE (A/O						
			MEFERENCE			25.	APPLICABLE		X
X PRIORITY NAVAIRS	SYSCOMREP PR	ICIA	0618187	Z OCT 67	AL.	unca 114	INCORPORATED		
H. YESNES			WEAPONS E	ENGINEERIN	IG DEPT	HRA D	11 Octo	ber	1961

ISASSEMBLY AND INSPECTION REPORT MAYNERS FORM 4730/2 (11-61)

REPORT SYMBOL BUMEPS 4730-2

26. FINDINGS:

- a. No. 1 blade was bent approximately 40 degrees aft at 14 inches from hub.
- b. Disassembly revealed that the two drive pins and two screws securing the blade bushing to No. 1 blade were sheared.
- c. Impression on teeth of No. 1 blade gear segment made by mating teeth of rotating cam indicated that propeller was in full low pitch position at time of impact. No indication of malfunction or material defects prior to impact was found.

10/5/67 KELLEY

OVERNAUL ACT	A STATE OF THE PARTY OF	n	ORT NO.	3. DATE OF B/1	4. ABBENOLY NO		MY 80.	·no	(20	00)	ENG IN
MAYAIR		Pensacola	796	10 Oct 67	D. DATE REMOVE	Prop.			MEMOVED FR	Ou (Bur	See 1
4G10-3		WH74055C		73030	UNK		20-86A	3.0	520721	1	,
TOTAL HES	12. HRS SINCE	13. BATE LAST	14. LAST DYERW		- Canal	IS. NO. PREV			17, A18	CRAFT (hamo)
UNK	UNK	UNK	UNK			HIMK.	T-28	3C	146	266	
OPERATING AC	TIVITY	19. FUR -EFR - AM	- 1/FH/GA	20. REASON FOR RE	EMOVAL AND CODE				-		
TRARON	FIVE	AAR 5-68	BA	Accident	/Incident	Dange	4.6.				
MO DISCRE	ANCY [B BASIC (MFG/DESI) DISCREPAN	GN) N	MON-BASIC (MAINT/OPER) DISCREPANCY	F	FOREIGN OBJ		Part He.)	T FAILURE	come.	Zone
DESCRIPTION	OF FINDINGS (In	clade asse and per	t so, of prinary	part failure)			22. 0	SCRÉPANT	PARTS (Part	80.)	ССВО
TO: B CC: N B CC CC S CC CC S CC CC CC CC CC CC CC CC	ATSF PHII ASC (AIR AS	4105B) 4041)	TA DISSON DE	R1820-20-6 AVAVNSAFECE AG AMSIX N.O. INAVMAT IRAFIP INCLANTFLT AS NORTON A RARON FIVE MARON FIVE	AFB (AVSAF O	WICER)		ERTINENT B MCOMPORTE MUMBE		VES	ETC.,
(See b	elow)				E DATE	7 2		_		-	-
RECOMENDAT	OHS	4		PRIO	KILI.	M					
None						10.00					
	MEGUESTED BY		905000	REPERENCE	consiste and		25. GRE	APP	LICABLE		X
PRIORITY	NAVAIRS	TSCOMREP P	NCLA	061818Z	OCT 67		114	1 1/100	DRPORATED		
SIGNATURE		1		SI. TITLE				32. 0	The same		
H. YESI	ES			WEAPONS I	ENGINE I	NG DEPT.	HEAD	11	Octob	er l	967
26. FI	DINGS:	ssible. D	nd damaged	to the ex	ection of			al che		the	4730

Governor is believed to have been operating satisfactorily prior to impact.

I. OVERHALE ACTIVITY ZAVAIREWORKFA	C Pensacola	798	10-11-67	4. ASSEMBLY N	DENCLITU (PART NO.		500)	X
R1820-86A	520721		66640	10-9-6	7	o room (Rag Hod)			
11. TOTAL HRS 18. HRS 5	19. DATE LAST	NAVAIREW	ORKFAC Pen	sacola	15. NO. PREV	T-280	101.5 - 1	46266	and the same
TRARON FIVE	AAR 5-6		Accident						
O NO DISCREPANCY	B X MFG/DES	IGN) N	MON-BASIC (MAINT/OPER) DISCREPANCY	F	FOREIGN OBJ	1240	7983	070	23
NASC (AI NASC (AI NAVAIRSI NASCIBCE	R 4113) R 5362) R 4041) SCOMREP PNCL REP WOODRIDG	CHNAVA A DIRAFI B CINCIA	NTREPO PAT AT P NTPLT	TERSON	-	Za. PERT	INERT BULLETINS.	CHANGES,	EYC
na va vn sa Cno	FECEN	CHATRA				INCO	NUMBER	YES	MO.
JAG 27. conclusions		TRAROI				R182	OEB607	X	
(See below)									
36. RECOMMENDATIONS	4	PI	RIOR	ITY					
None zs. PRIORITY NAVA	IRSYSCOMREP		REFERENCE	OCT 67		25. GRES 114	INCORPORATE	0	X
BO. SIGNATURE H. YESNES	9.	7	WEAPONS	ENGINEER	ING DEPI	HRA D	12 Oct	ber 1	1967

DISASSEMBLY AND INSPECTION REPORT MAYWEPS FORM 4730/2 (11-61)

REPORT SYMBOL BUMEPS 4730-2

- 26. FINDINGS:
- 1. Subject engine sustained severe fire and impact damage. The front crankcase was fractured and broken. All accessories and engine components aft of the power case were destroyed by fire.
- 2. Disassembly inspection revealed that the number seven articulated rod had fractured at the knuckle pin strap.
- 3. The number eight rod was fractured approximately five inches below the piston pin boss and the number six rod was slightly bent. The sides of both rods were damaged where contact was made with the fractured end of the number seven rod during operation.
- 4. The number seven cylinder skirt was broken on either side where contacted by the fractured number seven rod.
- 5. The 2045D19 crankcase main rear section to crankcase main front section bolt was broken between cylinders number seven and eight.

PRI DIR 798 12 October 1967 27. CONCLUSIONS: 1. None. 2. The number seven rod is considered to be the initial failure. A laboratory analysis of the rod failed to reveal the cause or origin of failure. The fracture surfaces were severely damaged from engine operation subsequent to the rod failure. 3. The damage incurred by the number six and eight rods was caused by the failure of the number seven rod. A laboratory analysis of the number eight rod revealed that it had failed from overload. 4.-5. Other internal damage resulted from the number seven art rod failure.

Code 015 22 NOV 1967

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES

THIRD ENDORSEMENT on TRARON FIVE, accident, ser 5-68A, concerning T-28C, BuNo 146266, of 5 Oct 1967, pilot KELLY

From: Chief of Naval Air Training

To: Commander, Naval Aviation Safety Center

Subj: Aircraft accident report; forwarding of

1. Forwarded, concurring in the conclusions and recommendations of the Aircraft Accident Board and comments by the subsequent endorsements.

ROBERT C. COATS
"Chief of Stage
(ting)

Philipal

Copy to: COMNAVAIRSYSCOM (AIR 404) CNABATRA NAVPLANTREPO, Columbus DIR, AFIP Washington, D. C. CO, TRARON FIVE

Code 015 1 4 NOV 1967 SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES SECOND ENDORSEMENT on TRARON FIVE AAR Ser 5-68A, concerning T-28C, BuNo 146266, occurring 5 October 1967, Pilot KELLY From: Chief of Naval Air Basic Training Commander, U. S. Naval Aviation Safety Center Via: Chief of Naval Air Training Subj: Aircraft Accident Report; forwarding of (a) CNABATRA Ltr Code 103 of 15 Mar 1967 1. Forwarded concurring with the conclusions and recommendations of the Aircraft Accident Board and the first endorsement, with the following additional comment: a. This command has previously recommended the installation of an ejection seat in the T-28 aircraft (Reference (a)). The latest estimate of possible ejection seat "saves" is 35 out of 61 fatalities in the Basic Training Command in the T-28 aircraft. 2. It is strongly recommended that re-consideration be given to this vitally needed installation at this time. Copy to: NAVAVNSAFCEN (2 direct) J. H. ARMSTRONG NAVAIRSYSCOM (AIR 404) Chief of Staff NAVPLANTREPO COLUMBUS DIR AFIP, WASH D. C. TRARON FIVE

ORIGINAL

Code 00 1 November 1967

SPECIAL HANDLING REQUIRED IN ACCORDANCE OPNAVINST P3750.6 series

FIRST ENDORSEMENT on TRARON FIVE AAR 5-68A, concerning T-28C, BuNo. 146266 accident occurring on 5 October 1967, Pilot ENS J. R. KELLY

From: Commanding Officer, Training Squadron FIVE
To: Commander, U. S. Naval Aviation Safety Center

Ha: (1) Chief of Naval Air Basic Training

(2) Chief of Naval Air Training

Subj: Training Squadron FIVE AAR 5-68A; forwarding of

Forwarded, concurring with the conclusions and recommendations
of the accident board, with the following comments:

a. If it can be said that there is a safety advantage for a Student Naval Aviator in the FCLP environment it lies in the relative simplicity of the required knowledge of emergency procedures. The students are constantly reminded of the need for rapid, positive action in the event of minimum altitude power loss: gear UP, canopy EMERCENCY OPEN, maintain flying speed, land straight ahead. (Hopefully, time permitting: switches, mixtures, fuel OFF).

(b) (5)

(b)(5)

b. Enclosure (4), statement of ENS (b) (6) requires amplifica-

(1) The two crash trucks in the hanger at Barin at the time of the crash were not there for use at Barin. This hanger is used as overnight storage for trucks that are assigned at outlying fields to the west and are parked at Barin to save the extensive roundtrip to Saufley.

(2) As discussed in the forwarding endorsement of TRABON
FINE AAR 3-66A the problem of rapid access to a crash scene is wellunderstood.

ORIGINAL

AIRCRAFT ACCIDENT REPORT
OPNAY FORM 3750-1A (Fee: 3-43) Prair 1

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AIRCRAFT ACCIDENT REPORT

SPECIAL HANDLING REQUIRED in accordance with

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TRAINING SQUADRON FIVE AAR 5-68A

PART V - THE ACCIDENT

Ensign James R. KBLLY took off from ALF Barin Field at 1650S, 5 October 1967 in T-28C BUNO 146266, side number 727, on a field carrier landing practice flight (CQ-10). Pre-flight, runup, launch and the next ten FCIP approaches were uneventful with no unusual tendencies noted by the LSO (Encl (1)). At approximately 1735 Ensign KELLY completed his upwind turn, dropped his speed brake, and transitioned to 82 kts, adjusting his distance abeam the runway to set up for his eleventh and final approach. At a proper distance abeam and about mid field the aircraft was heard backfiring and sputtering. Several witnesses stated that the aircraft was slowly descending and trailing black smoke. After the initial backfiring the engine appeared to pick up and run normally for a few seconds at which time it sputtered once more and then quit completely(Encl (2), (3), (4) and (5)). The aircraft was then observed in a level attitude settling into a cluster of sixty foot pine trees, 253 feet short of a cleared field. The aircraft initially hit the trees wings level, increasing its angle of bank to 30 degrees to the left just prior to impact (Encl (6) and (7)). As the aircraft impacted left wing down it flipped, rolled over on its starboard wing and came to rest inverted on a barbed wire fence at the perimeter of the cleared field (Encl (8) and (9)).

TRAINING SQUADRON FIVE AAR 5-684

PART VI - DAMAGE TO AIRCRAFT

The aircraft received ALFA damage as a result of the impact damage and ensuing fire.

The aircraft initially hit the trees wings level, increasing it's angle of bank to the left until it reached thirty degrees. After 106 feet of travel the port wing tip struck an 8 inch diameter tree 18 feet above ground leaving the wingtip suspended in the branches (Encl (7)). The aircraft then started skidding to the right as a result of the wingtip impact, with it's left wing still down. After 191 feet of travel the port main landing gear hit the ground followed shortly thereafter by the nosewheel which broke off on impact. The skidding and the sheared nose wheel evidently caused the aircraft to flip and roll over on the starboard wing as the port wing hit an 8 inch diameter tree just inboard of the outer wing panel. It was noted that just prior to the port main landing gear touchdown, the aircraft hit and broke off two 9 inch diameter trees in the vicinity of the starboard wing root, obviously weakening it sufficiently to allow the aircraft to roll over on it. After the aircraft became inverted it skidded another 31 feet, came to rest on a heading of 300° and burst into flames. The tail section was the only major component left reasonably intact. All other major components were damaged beyond repair as a result of the impact and subsequent fire.

TRAINING SQUADRON FIVE AAR 5-68A

PART VII - THE INVESTIGATION AND ANALYSIS

1. Personnel Factors

a. Pilot Factors

(1) The flight was an authorized syllabus hop at ALF Barin Field, Foley, Alabama for the purpose of field carrier landing practice. This was the second flight of the day for the student pilot.

(2) Investigation revealed no abnormal events concerning the pilot's personal activities that could have contributed to this accident (Encl (10)).

(3) Investigation of the wreckage revealed that the landing gear was down and locked and the speed brake was extended at the time of impact, despite the statement of one witness (Encl (5)). Had Ensign KELIX immediately raised the landing gear and speed brake on initial indication of engine failure, his glide ratio would have been increased sufficiently to execute a wheels-up landing in an open field. In addition, had he continued the slight right turn observed in enclosure (5), he would have avoided the group of trees which he subsequently encountered (Encl (6) and (7)). It is evident that the pilot's actions contributed substantially to the severity of the accident.

b. Maintenance, Servicing and Ground Handling Personnel Factors.

- (1) The investigation of all maintenance logs, records and procedures ascertained that the aircraft was ready for this flight (Encl (17)).
 - (2) The last calendar inspection was completed on 6 September 1967.
 - (3) Proper servicing and preflight requirements had been conducted.

c. Supervisory Factors

(1) The Board does not consider any supervisory factor as having contributed to this accident.

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TRAINING SQUADRON FIVE AAR 5-68A

PART VII - THE INVESTIGATION AND ANALYSIS (CONT'D)

2. Material Failure or Malfunctions

a. A priority DIR on Engine Model R-1820-86A serial number 520721
was requested by TRARON FIVE Message 060230Z OCT 1967. The results of the
DIR revealed that the number seven articulating rod fractured at the
knuckle pin strap. Number eight articulating rod fractured at mid point
and number six articulating rod was slightly bent and damaged. The
number seven rod was considered to be the initial failure, setting up the
chain of events which resulted in complete engine failure (Encl (12)).

3. Facilities

a. Facilities were not a factor in this accident.

4. NATOPS

a. Apparent failure of the pilot to retract his landing gear on initial indication of engine failure was not in accordance with recommended procedures outlined in the T-28 NATOPS Manual.

b. No change to NATOPS is recommended.

TRAINING SQUADRON FIVE AAR 5-68A

PART VIII - CONCLUSIONS

1. The Accident Board concludes that the primary cause of this accident was material failure of the number seven articulating rod of the engine resulting in complete loss of power.

TRAINING SQUADRON FIVE AAR 5-68A

PART IX - RECOMMENDATIONS

- 1. That continued emphasis be placed on a sound knowledge of emergency procedures with particular stress placed on the importance of retracting the landing gear and speed brake immediately when forced to ditch on unprepared terrain.
- 2. The nature of the mission of T-28 aircraft in the Naval Air Basic Training Command constantly exposes instructors and students to low altitude emergency situations. This situation is always present in Training Squadron FIVE operations in that aircraft are flown in the dirty configuration at a pattern altitude of 325° AGL, mostly over rough, unprepared terrain. Therefore it is recommended that suitable ejection seats be installed in T-28 aircraft in order to provide a better escape envelope for this type of operation.

VT-5 SERIAL 5-68A, 5 OCTOBER 1967, T-28C, BUNO 146266, PILOT KELLY

3. THE ENCLOSURES

- 1. STATEMENT OF LCDR (b) (6) (CONTROLLING 150)
- 2. STATEMENT OF AA (b) (6) (BARIN LINE DIVISION)
- 3. STATEMENT OF AA (b) (6) (BARIN LINE DIVISION)
- 4. STATEPENT OF ENS (b) (6) (STUDENT PILOT)
- 5. STATEMENT OF MR. (b) (6)
- 6. PHOTO SHOWING FLIGHT PATH AND POSITION OF WITHESSES
- 7. PHOTO SHOWLIG FLIGHT PATH THROUGH THE TREES
- 8. PHOTO SHOWING CENTERAL DAMAGE AND POSITION OF PILOT
- 9. PHOTO SHOWING GENERAL DAMAGE AND FOLDED STARBOARD WING
- 10. MEDICAL OFFICER'S REPORT
- 11. MAINTENANCE OFFICER'S STATEMENT
- 12. DISASSEMBLY AND INSPECTION REPORT (PRELIMINARY) WARF PHOLA 111934Z OCT 1967
- 13. RESCUE REPORT

STATEMENT OF LCDR (D) (6) USN, LSO, CONCERNING VI-5 MAR 5-68A

On 5 October 1967 I was controlling ISO for Flight 34 at AIF Barin Field.

On the first hop of the day (CQ-10) the flight had had shifting wind and rough air and had consequently flown a poor FCIP hop. ENS KELLY did not participate in the breakup and rendezvous portion of the hop due to radio failure in the turn up area, but had joined the flight for 6 passes of the FCLP portion of CQ-10. The entire flight was generally disheartened due to this poor performance so during the debrief I told them that if the second hop was in smooth air and flown well I would double the hop and adjust the grades of the previous hop.

Flight 34 launched at approximately 1700 on runway 21. The wind was from 210° to 190°, 5-6 kts, and the air was smooth. All phases of the launch and hop were normal and the entire flight was doing an above average job of flying FCLP. At approximately 1730 after 10 passes, I commenced calling the flight down. At this time I heard, "MAYDAY MAYDAY, this is 725 (garbled) aircraft ahead of me crashed into trees on fire or "aircraft ahead of me on fire crashed into trees." I hit the wave off light and turned to my right to search the pattern during this transmission. At this time, I noticed nothing unusual (I did not consciously count the aircraft). As I turned back toward the 180 position I noticed the black make starting to rise.

After transmitting crash, etc, and informing the fire house, I put the flight into Dolta at 10001. The crash crew and ambulance responded immediately, with (b) (6) AL, the LSC writer, riding the crash truck.

(b) (5). I then had the flight clean up in the Delta Fattern and orbit until the SAR helo could be diverted from the crash scene to act as crash safety equipment. At approximately 1805 I had the flight enter the break and make normal landings.

During the post flight debrief of the previous (CQ-10) hop, ESS NELLY

(b) (5)

The entire flight

showed normal concern due to their consistently below everage performance (failure to fly 82 kt attitude).

(b) (5)

SPECIAL HANDLING HE UIRED IN ACCORDANCE WITH OPNAVIRST P3750.6 SERIES ENCLOSURE (1) TO VI-5 AAR 5-68A

STATEMENT OF LCDR

(b) (6) usn iso. concerning vt-5 aar 5-68a (contid)

I have been a Naval Aviator since 1952, have approximately 4000 flight

hours and have been an ISO since 1954.

⁽⁴°(b) (5)

LCDR

2

STATEMENT OF AA (b) (6) CONCERNING VI-5 AAR 5-68A

When I first saw 727 he was due east of me. I saw at that time black amoke coming from the airplane and told AN (b) (6) who was walking up to the line shack with me from the west end of the line. About 3 seconds after sighting the smoke the engine started to sputter and die, then fire and run again for about 2 seconds, then quit altogether. He didn't move his flaps or hook or wheels. I didn't see the speed brake. He had his wings level and was (b) (5)

He kept losing altitude until I lost him in the trees. About 2 seconds later there was a big ball of fire and smoke where he had hit in the trees. (b) (5)

I still couldn't see the speed brake.

This is considered to be a credible statement.

STATEMENT OF AA (b) (6)

I saw aircraft 727 crash into what appeared to be trees.

(b) called my attention to it by saying "(b) (6) that bird is trailing smoke", at which time I looked toward the down wind leg of the FCLP pattern. I saw no smoke at this time but continued to watch the aircraft.

About 3 or 4 seconds later I heard what sounded like engine failure. Perhaps 2 seconds later the bird started losing altitude and when I saw that he wasn't going to gain altitude I knew a crash was more than likely. I didn't see if the gear were up or down. The aircraft maintained the same attitude as if he were flying a normal pattern. He maintained this attitude until I lost sight of him.

This is considered to be a credible statement.

STATEMENT OF ENS (5) (6) USH, STUDENT PILOT, CONCERNING VT-5 AAR 5-684

I received my commission through MROTO on 27 January 1967, and was immediately ordered to Pensacola for flight training. I have my private pilot's license and about forty civilian hours.

ENS KELLY and I went through Flight Prep together and flew most of our formation hope together at Whiting Field. (b) (5)

b) (5

On Thursday, 5 October 1967, ENS KELLY (b) (5)

(b) (5)

At

about 1700 we made a normal take-off and flew several FCLP approaches. On what was to be the final pass I was flying immediately behind EMS KELLY in the normal FCLP pattern and configuration. EMS KELLY was at about the abean position when I noticed that he was descending and streaming black smoke. The samks appeared to be coming from directly beneath the fuselage. At no time did I see EMS KELLY attempt to turn the aircraft. He flew straight ahead and crashed wings level into the trees just short of a farmer's field. From my position I was not able to tell whether or not his gear was up. The aircraft disappeared behind the trees and burst into flames upon impact. I then transmitted a may-day report to paddles.

I immediately cleaned up my aircraft and started orbiting Barin Field at 1000. The crash truck on station at the duty runway was able to reach the scene fairly rapidly because it was easily accessible by road. The SAR helicopter did not arrive on the scene until about twenty minutes later. The SAR helicopter was not on Barin frequency, so one of the pilots orbiting the field had to come up on Guard and tell the helo to change frequencies. That same pilot, ENS (5) then had to relay messages between paddles and the helo because they could not establish radio contact.

A second crash truck was needed at the scene of the accident to help put out the fire, and a crash truck was needed on the duty runway to recover those of us still orbiting the field. Although there were two trucks in the hangar at

SPECIAL HANDLING REQUIRED IN ACCOMMANCE WITH OPHAVIDAT 3750.6 MRIES ENCLOSURE (4) TO VT-5 AAR 5-661 STATEMENT OF ENS (b) (6) USN, STUDENT PILC

CONCERNING VT-5 AAR 5-684 CONT'D)

Barin, there was no one there to man them. The SAR helicopter had to leave the scene and stand by the duty runway so that we could land.



STATIMENT OF MR. (b) (6)

(b) (6)

CONCERNING VT-5 AAR 5-68A

On October 5, 1967 at about 5:45 PM, I was in my back yard and heard an altered backfiring and sputtering. I turned around and saw the aircraft in a slight right turn coming toward my barn. Initially his altitude looked arrmal as I constantly observe aircraft at Barin Field. At that time the inc picked up and sounded like it was running normally. Then the engine picked up and sounded like it was running normally. Then the engine it started to come up. A few seconds later the aircraft started settling into the trees in a near level attitude. For a second I lost sight of him would the trees and the next thing I saw was the flash fire. I then got into my pickup and drove to within 300 feet of the wreckage. I got out and ran directly toward the aircraft to see if I could see or help the pilot in any way.

b) (5)

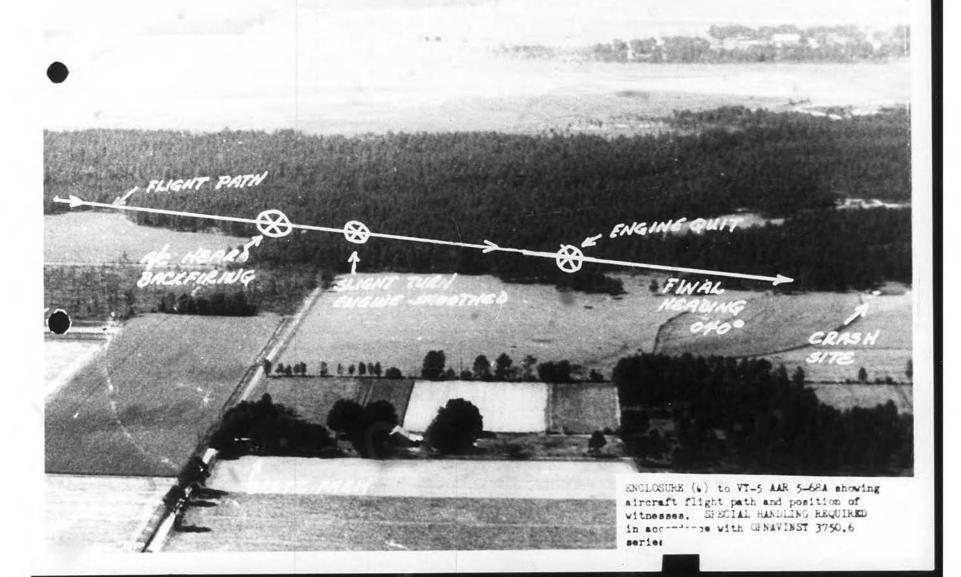
I have no previous aeronautical experience.

I am employed by NARF Pensacola as an aircraft preservation mechanic. I work with T-28, S2F, T-24, T-2B, F9F, HU-16, T-34, and occasionally a C-45.

I have worked at NARF Pensacola since August 8, 1955.

(b) (6

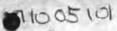
This is considered to be a credible statement.











REPORT OF POST-MORTEN BIOCHENICAL FINDINGS ON AIRCRAFT ACCIDENT FATALITY T-28C IN 6266 IDENTIFICATION Chief, Laboratory Service HS Heval Air Station KELLY, JAMES ROBERT 24/67 Pensacola, Florida WING /TESER US HAVAL AIR STATION PERSACULA PLA 30 OCT 67 M67-76 AP 88 PROSES T CO. USHASC, Horfolk, Van FIXED T. PHOTOS, KODA CHO OP-05F, Wash., D? C. I. AIRCRAFT ACCIDENT DATA: AIRCRAFT TYPE: AIRCRAFT SERIAL NUMBER: TOXICOLOGY: A. CARBON MONOXIDE BLOOD OR TIBBUE CONCENTRATION EXPRESSED AS CARROLYMENOGLOSIN SATURATION IS CARBOXYMENOGLOSIN SATURATIONS OF 10% OR ABOVE ARE CONSIDERED ELEVATED VALUES. B. LACTIC ACID THE CENTRAL HERVOUS SYSTEM LACTIC ACID CONCENTRATION IS . VALUES OVER 200 MANS ARE CONSIDERED TO BE INDICATIVE OF . STRAL MERVOUS SYSTEM HYPOXIA. THIS MAY BE DUE TO ONE OF SEVERAL CAUSES: (1) HIGH ALTITUDE EXPOSURE TO LOW OXYGEN TENSION; (2) REDUCED OXYGEN SUPPLY; (3) PROLONGED SHOCK; (4) TRAUMATIC INJURIES WHICH IMPAIR ADEQUATE GRYGENATION AND/OR CIRCULATION OF SLOOD TO THE CENTRAL MERYOUS SYSTEM. HON ELEVATED VALUES DO NOT RULE OUT HYPOXIA AS A CAUSE OF THE ACCIDENT C. ALCOHOL ETHYL ALCOHOL CONCENTRATION IS . . . D. DRUGS DETERMINATION FOR DRUGS OR POISONS WILL BE DONE ON REQUEST FROM THE BURNITTING PACILITY. REMARKS Condition of specimen: Good NOTE: PROZEM TIBLE VILL SE HELD FOR THENTY (20) DAYS. IF SPECIAL STUDIES OR PURTHER INFORMATION AND DESIRED. MANS. COMBULTATION BERVICE PAON THE ARROPACE BRANCH, AF(F, IS AVAILAGEON, D. C., RANDOLPH 3-1388 ON RANDOLPH 3-1000, EXTENSION 33. SUBMIT REQUEST BY MOST EN ABLE ON 24 HOUR BASIS. C EXAMINATION AND REPORT BY 5. APPROVED BY MEGE R. SMITH DATE OF REPORTS Captain, MC, 18

AFIP FORM (1 DET 66)

♦ C. S. SCHEDWEST PRINTING COVERS 1993—465174

1.	RS	(See code at right)			E - ESCAPE/EGRES	M - MAJ	FACTOR WEIGHT: M - MAJOR		
		A E		-	R S - SURVIVAL R - RESCUE	C+CON	TRIBUTING		
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17. E-RAY RESULTS									
17. E-RAY RESULTS	MODEL A/C	Leuno	-	100	I IDENTIFICATION OF PRINCIP	POAL			
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MEDICAL OFFICER'S REPORT OF A/C ACCURATE, INCIDENT, OR GROUND ACCIDENT - PAGE
OPMAY FORM 3750-88 (REV. 3-63)

BRECIAL HANDLING REQUIRED — See OPN. ST 1750-68 for Instructions.

SECTION E

INDIVIDUAL CHRONOLOGICAL DATA

SEE PAGE 8 PARA. 10 OF INSTRUCTION
TO BE COMPLETED ON PLANE COMMANDER, PILOT, CO-PILOT, OTHER INDIVIDUAL
IN CONTROL OF AIRCRAFT AT TIME OF MISHAP, AND/OR INDIVIDUAL CAUSING THE MISHAP

USE LOCAL TIME AND BRIEFLY RECORD ACTIVITY WITHIN EACH COLUMN

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1200	Muster	1200	Muster
1800	Secure		
OCT 67			
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1130	Secure		
Note:	No other personal chr-		
	onological information		
	available. Mrs. Kelly		
	left town shortly after	PHASE 1735	Material failure and
	the accident and this exam-		subsequent crash into trees on edge of field.
	iner had no opportunity	ESCAPE PHASE	
	to talk with her prior		
	to her departure. Close		
	friends were unable to		
	furnish further info-		
	rmation.		-13 - 12
		PHASE	
		100	

TIME OF RESCUE IDENTIFICATION OF INDIVIDUAL MODEL N/C 146266 6-68 T-280 SILA

NAME OF INDIVIDUAL

KELLY, James Robert Ens USNR

ECTION !	Name and Address of the Owner, where the Owner, which is	430		PATI	HOLO	GICAL I			(Refer in Section F of instructions.
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		- Section 1							III FORE OF HOSPITALISM
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and the same of th	TES YES								
IS, PRIMARY CAUSE OF DEATH							CONDART	CAUSE OF DEATH	4
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MEDICAL OFFICER'S REPORT OF A CICIDENT, INCIDENT, OR GROUND ACCIDED - PAGE 4A OPHAY REPORT STSO.7

OPNAY FORM 3750-8D (REV. 3-83)

SPECIAL HANDLING REQUIRED, See OPNAY INST \$750.5E for instructions

SECTION F (Continued)

SURFACE INJURIES

DESCRIBE AND SHOW GRAPHICALLY BY OUTLINING AND SHADING AFFECTED AREAS ALL LACERATIONS, ABRASIONS, CONTUSIONS, PUNCTURE WOUNDS, SPRAINS AND BURNS

RECORD ALL INJURIES NO MATTER HOW TRIVIAL, WHETHER PATIENT LIVED OR DIED

6-68 T-280

146266

IDENTIFICATION OF INDIVIDUAL SNA

HAME OF INDIVIDUAL KELLY, James Robert

ENS USNR

MEDICAL OFFICER'S REPORT OF A/C ACCOUNT, INCIDENT, OR GROUND ACCIDENT - PA OPNAY FORM 3730-8E (Res. 3-63) SPECIAL HANDLING REQUIRED See OPNAYINST 3750-8E for in-SPHAY REPORT 3750-7 SECTION F (Continued) SKELETAL INJURIES DESCRIBE AND SHOW GRAPHICALLY BY OUTLINING
ALL FRACTURES BY TYPE (Simple, compound, comminuted, etc.) AND DISLOCATIONS INDICATING DIRECTION OF DISPLACEMENT. DESCRIBE AND SHOW GRAPHICALLY: I. ALL FRACTURES OF SPINAL COLUMN (Simple, compressed, etc.)
2. DISLOGATION AND DIRECTION OF DISPLACEMENT. 1. SITES OF CORD DAMAGE, IF ANY. DETAILS OF SPINAL INJURIES MODEL A/L IDENTIFICATION OF INDIVIDUAL 6-68 T-280 146266 SNA KELLY, James Robert

MEDICAL OFFICER'S REPORT OF A CCIDENT, INCIDENT, OR GROUND ACCURIT - PAGE 5 OPHAN REPORT STOOD OPHAN FORM STROOD (REV. 3-63)

SPECIAL HANDLING REQUIRED. See OPHAN ENST \$150.6E for Instructions
SECTION G

ESCAPE, PERSONAL AND SURVIVAL EQUIPMENT

OPNAY PORM 2780-8F (REV. 2-63) SECTION G

LIST AND CODE IN ACCORD	ANCE WITH SECTIO	G OF IN	STRUCTI	ONI	PHASE	10007-0000000	-ACCIDENT/MISHAP	E-ESCAPE/EGREES PHASE
EQUIPMENT DESCRIPTION INCLUDING SPECIFIC MODEL DESIGNATION	MODIFICATION	RE. QUIRED	AVAIL-	S. NEED	6. USED	7. FAILED	(Explain failures, loss,	MARKS , and/or difficulty encoun- 1952 plain paper if needed.)
Shoulder Harness (MS 16069/1) Ap Belt (22033/1) Inertia Reel (MH-1) Plight Suit, Summer Tan Helmet (APH-6A) Visor, Tinted Field Shoe (M-1) Ploves, Summer Inife, Survival Pirst Aid Kit Parachute (NB-6) Aife Jacket (NK-2) Aiferaft (PK-2)	None None None None None None None None	***************************************	A A A A A A A A A	A A A A A A A	A A A A A A			be determined eckage debris
4.7		1						

NARRATIVE OF ESCAPE/EGRESS, SURVIVAL AND RESCUE PHASES

MOR HO. 6-68

SECTION H

HODEL A/E T-280

146266

IDENTIFICATION OF INDIVIDUAL

SNA

	11			VE/EGRESS/SURVIVA	AL PHASES	HEFER TO MECTION I OF INSTRUCT	TIONS
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-		4. ATTEMPTED	100				
		B. ACCOMPLISHED					
		6. THRU CANOPY					
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	X	8. DURING EGRESS					
	X	9. SUBSEQUENT TO EC					12. SEQUENCE OF EJECTION
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P4	lot	in aircraft beh	ind M	r. KELLY sav			
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200	101000000	P LOCATING ACCIDENT SITE			3.		
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Distriction of the control of the co	O INDITION OF THE LAND OF THE	VIDUAL DEPART FROM LAND Explain reason and sequence YES ASE TRAINING MAR 1967 TO ACK OF TRAINING AND/OR TYPES Mr. KELL	CTION VER	29 MAR 196	TRAINING F/ 7 SEAT ANY PHASE also his	COORS TION OF THIS SIBHAPS (U yes, explain	Maria Comment

ENCLOSURE (1)

USNAAS SAUFLEY FIELD, PENSACOLA, FLORIDA MOR #6-68
OCCURRING 5 OCTOBER 1967, CONCERNING JAMES ROBERT
KELLY, (D) (6) ENSIGN USNR

CONCLUSIONS AND RECOMMENDATIONS

Mitnesses state that they heard Mr. KELLY'S aircraft sputter and backfire about midfield in the pattern and then commence to trail black smoke and descend. The aircraft was then noted to "catch on" and run smoothly for a few seconds before it began again to backfire and then quit. Mr. KELLY was seen to continue his descent straight ahead with wings level until his port wing hit several pine trees which put the aircraft into a left skid with a 30° angle of bank to the left. Then the port wing is thought to have struck the base of a big pine tree with simultaneous collapse of the nose gear. With this collapse the engine dug into the ground and flipped the aircraft inverted to the right and folded the starboard wing beneath the inverted functor. The aircraft burst into flames immediately thus (b) (6)

It is postulated by the Accident Board that if Mr. KELLY had raised his landing gear in accordance with emergency procedures, he probably would have cleared enough of the trees to allow him to make a landing in the adjacent open field.

(b) (5)

In conclusion, it is recommended that continued emphasis be placed on the necessity for all pilots to be unquestionably, thoroughly, and completely familiar with all emergency procedures. Furthermore, it is strongly recommended that a study be instituted to determine the feasibility of installing ejection seats in the T-28C. In this socident and many similar in the past, (b) (5)

STATEMENT OF LCDR (b) (6) USNR, MAINTENANCE OFFICER, TRARON FIVE CONCERNING VT-5 AAR 5-684.

1. The results of the priority DIR on the Engine Model R-1820-864 Ser. No. 520721 revealed the number seven articulating red fractured at the knuckle pin strap and the cylinder skirt. Broken number eight articulating red fractured at mid point and number six red was slightly bent and damaged.

Number seven red is considered to be the initial failure. Reference
NARF MSG. 11 1934Z CCT. 1967.

2. A review of the history of the engine indicates that prior to the accident the material and mechanical conditions were satisfactory. Immediately preceding the crash the aircraft had flown 12 consecutive flights for a total of 6.8 hours without any discrepancies being reported.

(b) (6) LGDR USNR NAMNZOZO SAU 390

VY-5

RITUZYUW FUCLMHARAD SEMIDET-DUUM--BUCILGA.

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) IIIDSAZ OCT GT

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IO ZENZMAVAIRSYSCOMEP PNCLA

LENZTRARON FIVE
INFO ZENZONATRA

ZENZONABATRA

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"HI DIR ON RIBES-BSA ENG SER SPSYS!

A. NAVAIRSYSCOMREP PNGLA SESSES DCT ST

S. NAVAIRSYSCOMREP PNGLA CONTROL NE RISSD-20-68

C. TRARON FIVE AAR S-SSA

J. NAVAIRINST ATID.6

L. ENGINE SUSTAINED SEVERE IMPACT AND FIRE DAMAGE

FRONT CRANKCASE FRACTURED AND BROKEN. ALL ACCESSORIES AND ENGINE COMPONENTS AFT OF POWER CASE DESTROYED BY FIRE.

2. DISASSEMBLY INSPECTION REVEALED NUMBER SEVEN ART ROD FRACTURED AT KNUCKLE FIN STRAP AND CYLINDER SKIRT BROKEN. NUMBER EIGHT ART ROD FRACTURED AT MID POINT AND NUMBER SIX

PAGE TWO RUCLMHAB405 UNCLAS
SLIGHTLY SENT AND DAMAGED.

3. DISASSEMBLY OF PROP AND PROP GOV REVEALED NO CONTRIBUTING FACTORS. CARB TOO BADLY DAMAGED TO CHECK.

4. ANALYSIS OF NUMBER SEVEN ROD COULD NOT ESTABLISH CAUSE OF FAILURE. NUMBER EIGHT ROD RAILED FROM OVERLOAD. NUMBER SEVEN CONSIDERED TO BE INITIAL FAILURE.

5. DIRS FOLLOW.

TOR225/Z 11 OUT 67/V3 LNR26 111934Z

SPENIAL HANDLING REQUIRED IN ACCORDANCE WITH OPPRVINGT P3750.6 SERIES ENGLOSURE (12) TO VT-5 AAR S-68A

CHRONOLOGICAL SEQUENCE OF EVENTS

051745	RCC received alert via Basic Training Command telephone extension 3222 from Mrs. (b) (6) of Alberta, Alabama who reported that an aircraft had just crashed near her home. Location of crash site one mile west and one mile
	south of Alberta. Aircraft is on fire.
051746	UH-3LT 145667 (Pedro) airborne proceeding to crash site.
051747	RCC received above crash information from NAAS Sauriey Field via crash net 30-GP-1086. Aircraft identified as T-28
051755	28727. RCC notified Basic Safety of crash.
051756	Pedro on deck at crash site. One crash truck at scene;
0)11/0	aircraft still burning. Doctor and crewmen out of Pedro to assist crash crew.
051800	Pedro airborne crash site proceeding to Barin Field.
051803	Pedro on deck Barin Field standing by for ISO to land remainder of flight.
051810	Pedro airborne Barin Field to direct additional crash equipment to scene.
051811	Pedro reported to RCC that aircraft crashed inverted and at present is still burning. Unconfirmed Code Alpha. Basic Safety notified.
051816	Pedro on deck at crash sits.
051832	Pedro airborne proceeding to Barin Field to pick up VT-5
-,	Safety Officer.
051835	Pedro on deck Barin Field.
051837	Pedro airborne Earin Field to take Safety Officer and direct additional crash equipment to site. Aircraft at this time still burning. Pilot confirmed Code lpha.
051845	Pedro on deck crash site.
051855	UH-2B 151328 returning from NAAS Whiting Field diverted to scene for Airborne radio relay.
051858	Pedro 328 on scene crash site.
051859	Pedro 328 relayed information from RCC that either the doctor or corpsman must go to Foley, Alabama for the coroners
051926	Pedro 667 airborne crash site. 328 and 667 departing scene returning NAS Fensacola. Doctor left on scene.
051940	Pedro 328 and 667 on deck NAS Pensacola. Case closed.

UH-34J BUNO 145667 CREW LIST

Pilot:

LT (b) (6

Medical:

LT

Crew:

AMH3 ATN2

UH-2B BUNO 151328 CREW LIST

Pilot:

 $_{\pi}$ (b) (6)

Crew:

ADJ2 (b) (6) ADR2

NNNNZCZCNASC 885RZCZCNASC 886 RITUZYUW RUCLMHA Ø405 2841927-UUUU--RUCILSA . ZNR UUUUU R 1119347 OCT 67 FM NAVAIREWORKFAC PNCLA TO ZEN/NAVAIRSYSCOMREP PNCLA ZEN/TRARON FIVE INFO ZEN/CNATRA ZEN/CNABATRA RUCILSA/NAVAVNSAFECEN BT UNCLAS PRI DIR ON R1820-86A ENG SER 520721 A. NAVAIRSYSCOMREP PNCLA 061818Z OCT 67 B. NAVAIRSYSCOMREP PNCLA CONTROL NR R1820-20-68 C. TRARON FIVE AAR 5-68A

D. NAVAIRINST 4730.6

885 67

Cog: M&M

FRONT GRANKCASE FRACTURED AND BROKEN. ALL ACCESSORIES
AND ENGINE COMPONENTS AFT OF POWER CASE DESTROYED BY FIRE.
2. DISASSEMBLY INSPECTION REVEALED MUBBER SEVEN ART ROD
FRACTURED AT KNUCKLE PIN STRAP AND CYLINDER SKIRT BROKEN.
NUMBER EIGHT ART ROD FRACTURED AT MED POINT AND NUMBER SIX

I. ENGINE SUSTAINED SEVERE IMPACT AND FIRE DAMAGE.

PAGE TWO RUCLMHA 0405 UNCLAS
SLIGHTLY BENT AND DAMAGED.
3. DISASSEMBLY OF PROP AND PROP GOV REVEALED NO CONTRIBUTING
FACTORS. CARB TOO BADLY DAMAGED TO CHECK.
4. ANALYSIS OF NUMBER SEVEN ROD COULD NOT ESTABLISH CAUSE OF
FAILURE. NUMBER EIGHT ROD RAILED FROM OVERLOAD. NUMBER SEVEN
CONSIDERED TO BE INITIAL FAILURE.
5. DIRS FOLLOW.
BT

OCT 111934Z

T. 280 /146266

NNNNZCZCNASC263CZCSLB 058 PTTEZYUW RUCLMHA7769 2790230-EEEE--RUCILSA. ZNS EEEEE P 060230Z OCT 67 FM TRARON FIVE TO RUENAAA/CNO RUCILSA/NAVAVNSAFCEN ZEN/NAVAIRSYSCOMREP PNCLA INFO RUEDBHB/NAVAIRSYSCOMHQ RUCIJFA/JAG RUCLBEA/COMSIX ZEN/CNATRA ZEN/CNABATRA RUEDFIR/NAVPLANTREPO PATTERSON RUEDBHB/CHNAVMAT RUEOPDA/DIR AFIP RUEDNKA/CINCLANTFLT RUWJABA/DAS NORTON AFB BT UNCLAS E F T O FOR OFFICIAL USE ONLY PPELIMINARY/SUPPLEMETARY MESSAGE REPORT OF AIRCRAFT ACCIDENT A. OPNAVINST 3750.6F

PAGE TWO RUCLMHA 7769 UNCLAS E F T O 1. 5 OCTOBER 1967, 1735S, DAY 2. T-28C, BUNO 146266, VT-5, 5-68A 3. 1/2 MILE NE OF ALF BARIN FIELD, FOLEY, ALABAMA 4. JAMES ROBERT KELLY, ENSIGN, (b) (6), USNR, 1395, ACTIVE, ALFA 5. NONE 6. ALFA 7. ID3 . 0.7 B. ATTEMPTED CONTROLLED DITCHING (LAND) 9. AT THE 180 DEGREE POSITION OF FCLP PATTERN STUDENT PILOT APPARENTLY HAD AN ENGINE FAILURE . BOLACK SMOKE WAS SEEN COMING FROM THE AIRCRAFT WHILE IT WAS STILL AIRBORNE, SEVERAL WITNESS HEARD THE ENGINE CUTTING IN AND OUT. THE A/C HIT 60 FT TREES APPROXIMATELY 500 FT SHORT OF A CLEARED FIELD. THE A/C DROVE THROUGH THE TREES. FLIPPING OVER AND COMING TO REST INVERTED ON THE EDGE OF THE PLOWED 10. W/V 010/5KTS RELATIVE TO RWY, CLEAR, TEMP 80, DEW PT 60, VIS 8NH 11 . NONE 12. REQUEST PRI DIR ON R-1829-86A ENG SER BL 520721 AND ACCESSORIES 13. NA

PAGE THREE RUCLMHA 7769 UNCLAS E F T O

14 . NA

15. SEVERAL PINE TREE KNOCKED OVER IN AIRCRAFT PATH OF TRAVEL.

6. (b) (6) LT., ASO, AUTOVON 899-3358 EXT 98-247

T28C 146266